

Use of Expert Judgment  
in Risk Assessments  
Involving Complex State Spaces

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# MOTIVATION

- Detailed inspections of in-service wiring show that problems are common to both large and small transport aircraft:
  - ◆ inadvertent damage during maintenance, such as using wire bundles as ladder rungs, stepping on and damaging wiring hidden under insulation blankets,
  - ◆ inadequate support clamping,
  - ◆ improper installation that can aggravate chafing
- Today's jet aircraft rely more and more on sophisticated electrical and computer systems, placing a premium on the reliability of wiring, power feeder cables, connectors and circuit protection devices.

# MOTIVATION

- The physical failure of wiring has
  - ◆ caused damage to other aircraft systems
  - ◆ ignited flammable material in close proximity to wiring.
  - ◆ caused malfunctions that have contributed to turnbacks and in-flight diversions
- The amount of wiring in transport category aircraft has grown steadily over time, with no plateau yet visible. The more of it, the greater the potential exposure to wiring failures.

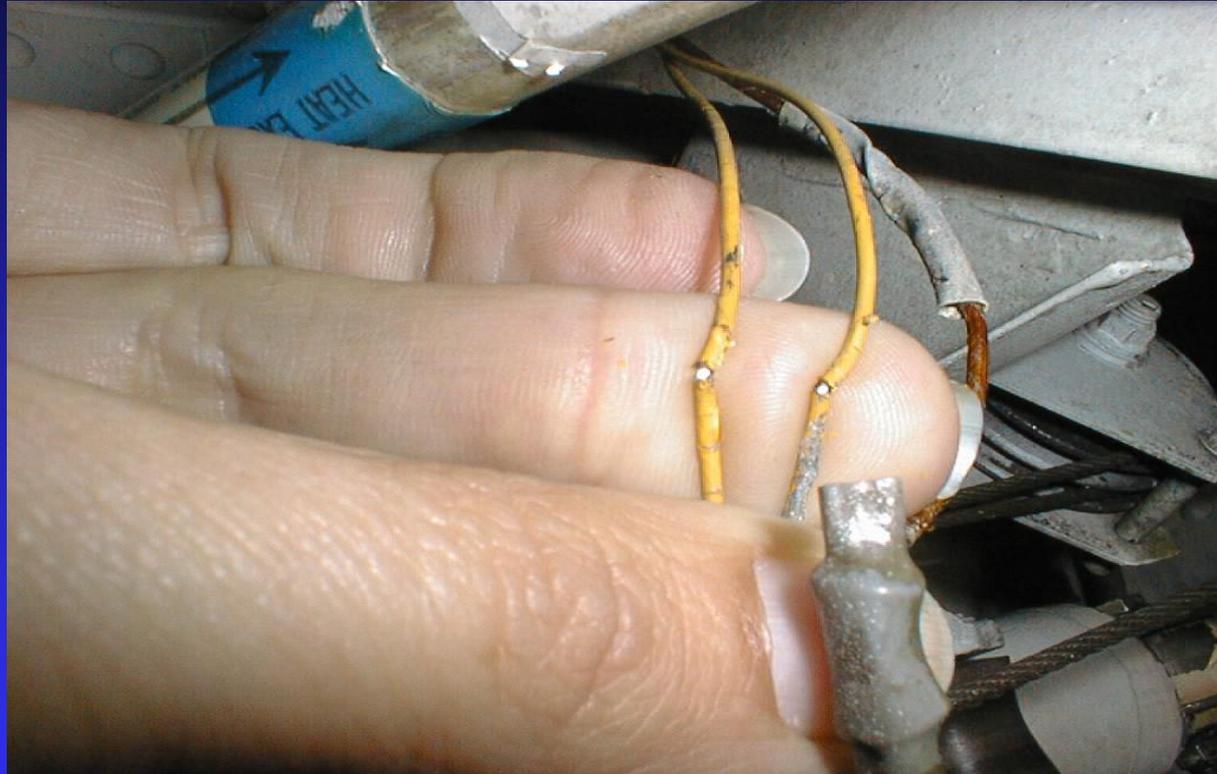
## MOTIVATION

*“The increasing reliance on electrical power on modern and future public transport aircraft for flying control, engine and flight management systems with the associated increase in the use of computers, in addition to passenger services and entertainment systems, makes such aircraft more vulnerable to electrical fires and their potential effects, particularly if the flight crew do not receive timely warnings of electrical fire initiation.”*

(Investigative report United B767-300 on a Jan. 9, 1998, the UK’s Air Accidents Investigation Branch)

Wires failures events can occur at three levels

# Wire Level



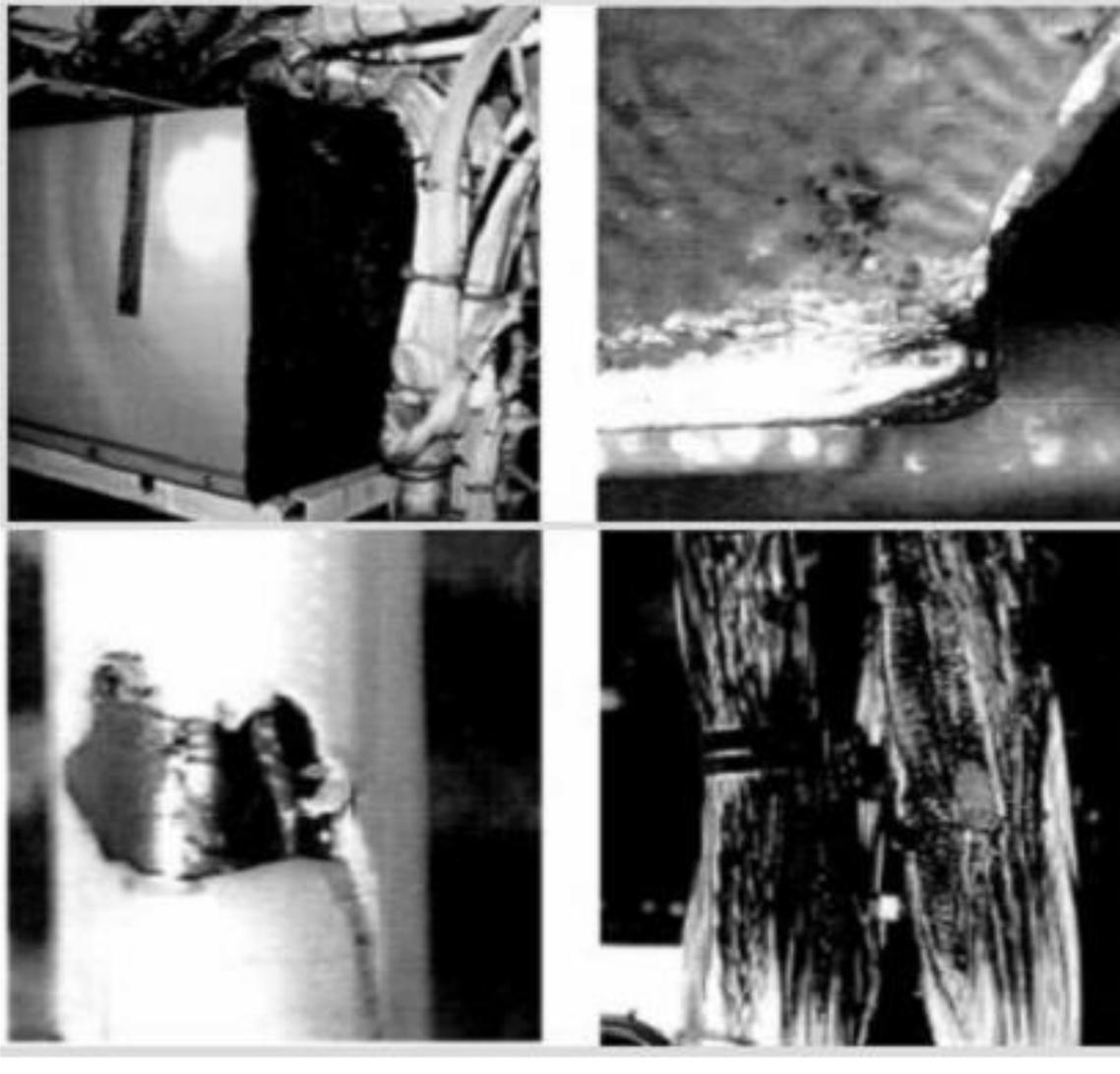
Insulation has faults. An EWIS failure probably has not occurred yet but the probability of an EWIS event is much higher. A common cause fault is indicated as the breach in the insulation line up.

# Bundle Level



An arcing event has occurred. It is assumed that the arcing event began with one or two wire chaffing against the standoff. However, as a result of the arcing many wires in the bundle have failed. The possible effect of the failure depend on which systems are routed in the bundle.

# Zonal Level



Upper Left: Install chiller in EE bay. Large object in a zone with high wire density.

Upper Right: Rough metal edge of cooler.

Lower Left: Chafed wire.

Lower right: Resulting arcing in two adjacent bundles.

United Airlines B767-300, Jan. 9, 1998

# DEVELOPEMENT OF WIRE FAILURE MODEL

## ■ Failure Modes

- Opens: “fail to open”
- Shorts: “fail to ground”

## ■ Failure Density

$$f(t_i|\lambda_i) = \lambda_i \exp\{-\lambda_i t_i\} \quad \text{where } i=o, g$$

## ■ Time until wire failure

$$T = \text{Min}\{T_o, T_g\} \sim \exp(\lambda_o + \lambda_g)$$

- To completely specify the distribution, this parameter must be estimated, usually from past data

# INCORPORATION OF ENVIRONMENTAL VARIABLES

- But there are many types of wiring environments and these environments will affect the failure rates
- A common model for incorporating the affect of covariates is the proportional hazards model (PHM)
- The basic idea of the model is to write the failure rate as a function of the covariates  $X_1, \dots, X_n$

$$f(t|\beta_0, \beta_1, \dots, \beta_n) \\ = [\exp\{\beta_0 + \sum_{j=1,n} \beta_j X_j\}] \exp\{ -[\exp\{\beta_0 + \sum_{j=1,n} \beta_j X_j\}]t \}$$

where  $\beta_0$  is some base failure rate and  $\beta_i$  reflects the influence of  $X_i$  on the failure rate

- ◆ but not much failure data exists except for a few wire types

# EXPERT JUDGEMENT USING PAIRED COMPARISON

- Paired Comparison
  - ◆ Designed to measure group preferences for a set of objects by letting subjects judge the objects 2 at a time
    - ◆ for each pair of objects, each subject simply states which of the 2 objects (s)he prefers
  - ◆ Allows for statistical tests for
    - ◆ individual expert responses
    - ◆ expert responses as a group
- Models for paired comparison
  - ◆ Thurstone (1927)
  - ◆ (Bradley and Terry, 1953)
    - ◆ These models also provide goodness of fit tests

# OVERVIEW

## PAIRED COMPARISON

- Set up
  - ◆ Let  $E_1, \dots, E_n$  denote the objects to compare
  - ◆  $e$  experts are asked a series (specifically a total of  $n$  taken 2 at a time) of paired comparisons as to which they prefer – the idea is that comparing items two at a time is easier than comparing items all at once
  - ◆ Let  $N_r(i)$  represent the number of times that expert  $r$  preferred  $E_i$  to any other
  - ◆ The paired comparison results yield values  $N_r(1), \dots, N_r(n)$  for each expert  $r = 1, \dots, e$ .

# OVERVIEW

## PAIRED COMPARISON

- Testing if each expert is specifying a true preference structure in his/her answers or just assigning answers in a random fashion.
  - ◆ This can be determined by analyzing the number of circular triads in his/her comparisons.
 
$$E_1 > E_2, E_2 > E_3, \text{ and } E_3 > E_1$$
  - ◆ David (1963) determined that  $c(r)$ , the number of circular triads in expert  $r$ 's preferences, is given by

$$c(r) = \frac{n(n^2 - 1)}{24} - \frac{1}{2} \sum_{i=1}^n \left( N_r(i) - \frac{1}{2}(n-1) \right)^2$$

# OVERVIEW

## PAIRED COMPARISON

- Kendall (1962) developed tables of the probability that certain values of  $c(r)$  are exceeded under the null hypothesis that the expert answered in a random fashion for  $n = 2, \dots, 10$ .
- In addition, Kendall (1962) developed the following statistic for comparing  $n > 7$  items

$$c'(r) = \frac{n(n-1)(n-2)}{(n-4)^2} + \left( \frac{8}{n-1} \right) \left[ \left( \frac{1}{4} \right) \binom{n}{3} - c(r) + \frac{1}{2} \right]$$

- The above is chi squared with  $n(n-1)/(n+2)$  df
- Expert eliminated if we the random preference hypothesis cannot be rejected at the 5% level of significance

# OVERVIEW

## PAIRED COMPARISON

- The agreement of the experts as a group can be statistically validated. Let  $N(i,j)$  denote the number of times some expert preferred  $E_i$  to  $E_j$ .
- To test the hypothesis that all agreements of experts are due to chance, Kendall (1962) defines the *coefficient of agreement* as

$$u = \frac{2 \sum_{i=1}^n \sum_{j=1, j \neq i}^n \binom{N(i,j)}{2}}{\binom{e}{2} \binom{n}{2}} - 1$$

# OVERVIEW

## PAIRED COMPARISON

- Kendall tabulated distributions of 
$$\sum_{i=1}^n \sum_{j=1, j \neq i}^n \binom{N(i, j)}{2}$$

for small values of  $n$  and  $e$  under the hypothesis that all agreements of the experts are due to chance.

- For large values of  $n$  and  $e$ , Kendall (1962) developed the statistic

$$u' = \frac{4 \left[ \sum_{i=1}^n \sum_{j=1, j \neq i}^n \binom{N(i, j)}{2} - \binom{e}{2} \binom{n}{2} \left( \frac{e-3}{2} \right) / (e-2) \right]}{e-2}$$

which is chi-squared,  $df = n!e(e-1)/[2!(n-2)!(e-2)^2]$

- The hypothesis that all agreements are due to chance should be rejected at the 5% level of significance

# OVERVIEW

## BRADLEY-TERRY MODEL

- Assumes that the true “value” of object  $i$  is  $h_i$
- If experts can be treated as independent samples for each question then the probability that object  $i$  is preferred to object  $j$  is expressed as  $p_{ij} = h_i / (h_i + h_j)$
- Given that  $i$  and  $j$  are compared  $e$  times, the probability of seeing  $i$  preferred to  $j$  exactly  $N(i,j)$  times,  $i, j = 1, \dots, k, i < j$ ; is

$$L = \prod_{i < j} \binom{e}{N(i, j)} p_{ij}^{N(i, j)} (1 - p_{ij})^{e - N(i, j)} = \prod_{i < j} \binom{e}{N(i, j)} \left( \frac{h_i}{h_i + h_j} \right)^{N(i, j)} \left( \frac{h_j}{h_i + h_j} \right)^{e - N(i, j)}$$

- Find  $h_i$  through maximum likelihood estimation

# OVERVIEW

## BRADLEY-TERRY MODEL

- Note that the values can be determined up to a constant, that is if  $h_i$  are solutions so are  $Ch_i$
- Ford (1957): The following iterative solution procedure can be used to solve for the  $h_i$  up to a scale constant **provided that it is not possible to separate the  $n$  objects into two sets where all experts deem that no object in the first set is more preferable than any object in the second set.** Letting  $N(i)$  denote the number of times some expert prefers  $E_i$  over any other item

$$h_i^{(k+1)} = \frac{N(i) / e}{\sum_{j=1}^{i-1} \left[ h_i^{(k)} + h_j^{(k+1)} \right] + \sum_{j=i+1}^n \left[ h_i^{(k)} + h_j^{(k)} \right]}$$

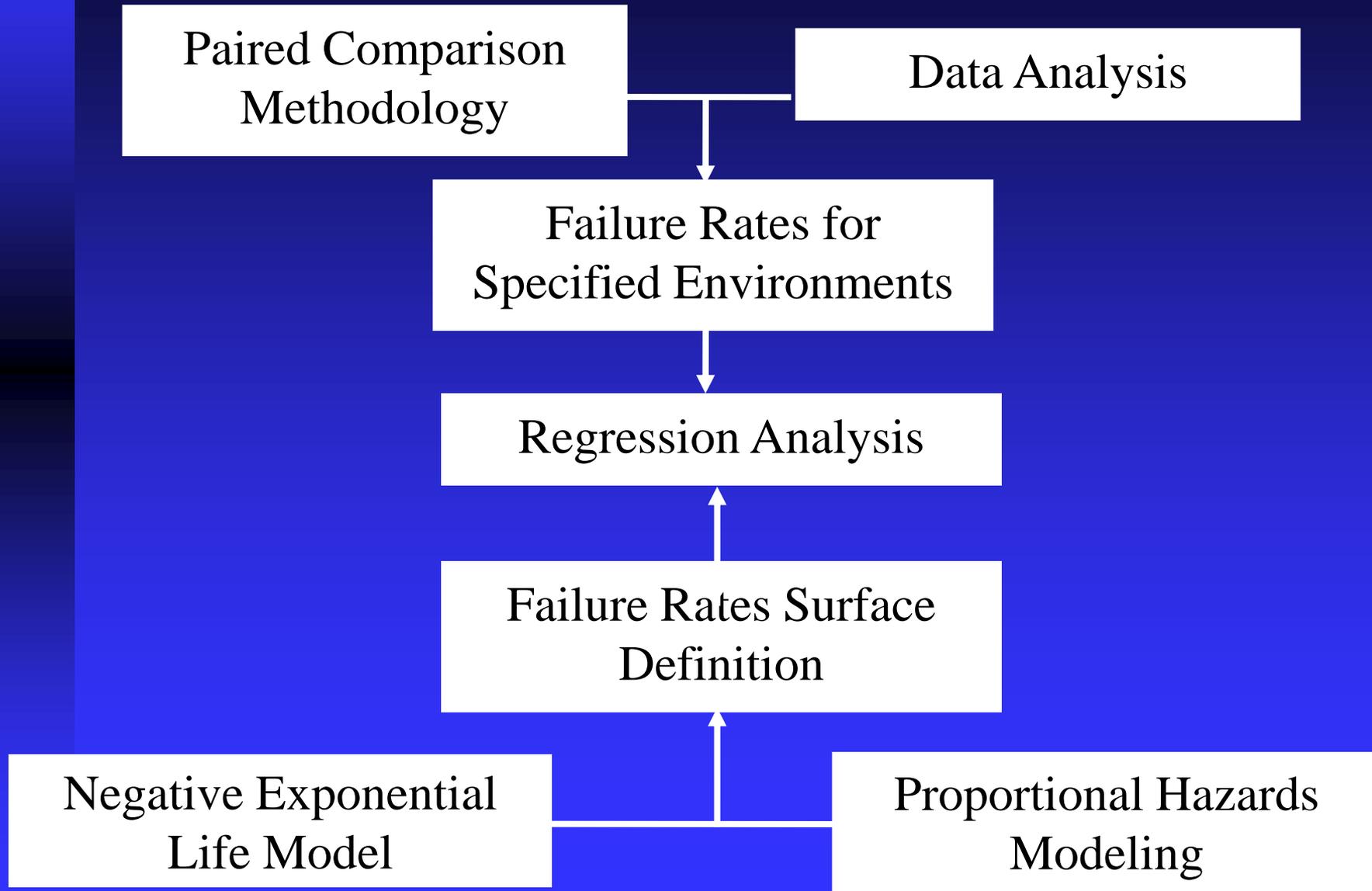
# OVERVIEW

## NEL MODEL Cooke (1991)

- But the Bradley-Terry Model is for probabilities not failure rates!
- Note if  $T_i \sim \exp(\lambda_i)$  then  $\Pr\{T_i < T_j\} = \lambda_i / (\lambda_i + \lambda_j)$
- Thus instead of asking experts “which object do you prefer”, we can ask “given two environments which environment will produce a failure first” and use all the paired comparison and Bradley-Terry Methodology
- Given that the values  $h_1, \dots, h_n$  are failure rates obtained to within a scale constant, if we can, from another method, determine an exact estimate of one of the failure rates, say  $h_j^+$ , we may calculate estimates as

$$h_i^+ = (h_j^+ / h_j) * h_i \quad i=1, \dots, n$$

# OVERVIEW OF APPROACH



# DEFINE THE ENVIRONMENT: ENVIRONMENTAL VARIABLES

Variables	Levels			
	1	2	3	4
Wire Gauge	4\0-8 awg	10-16 awg	18-22 awg	24-26 awg
Conductor Type	Aluminum	Copper	High Streng. Copper Alloy	
Splices	None	Environmental	Non-environmental	
Bundle Protection	Some Level of Protection	Not Protected (Open)	Protected Metal Conduit	
Curvature of Bundle	Low (> 10x)	High (<= 10x)		
Ops/Main Traffic	Low	Moderate	High	
Vibration	Low	Moderate	High	Extreme
Ops temp\presurization	Benign (P&T Controlled)	D1- P Contrl. but not T	D2 (P&T not controlled)	D3 (High T, P not contrl)
Exp Corrosive Fluid	No	Yes		
Exp Conducting Fluid	No	Yes		
Bundle Size	Large (> 1.25 in)	Moderate (0.5-1.25 in)	Small (0.2-0.5 in)	Very Small (< 0.2 in)
Insulation Type	Polyimide	Hybrid (PI/FP Composite)	ETFE & other FPs	
Bundle Orientation (Shock)	Horizontal/Vertical Wire	Longitudinal		

- an upper bound of  $4 * 3 * 3 * \dots * 2 = 995,328$  environments

Note that these are categorical

# QUANTIFY VARIABLES

- Define environments via explanatory variables

## EFFECT OF SINGLE VARIABLES ON OPEN FAILURES

Page 2

### BUNDLE PROPERTIES

#### Bundle Size

Large (> 1.25 in)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9
	less severe <-----> more severe																
Moderate (0.5-1.25 in)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9
Small (0.2-0.5 in)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9
Very Small (< 0.2 in)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9

#### Bundle Protection

Some Level of Prot.	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9
	less severe <-----> more severe																
Not Protected (Open)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9

#### Curvature of Bundle

Low (> 10x)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9
	less severe <-----> more severe																
High (<= 10x)	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9

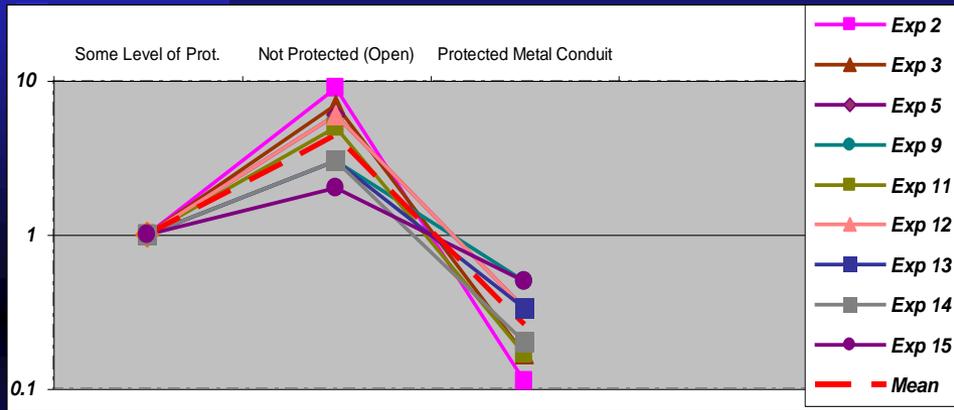
#### Bundle Orientation (Shock)

Horizontal/Vertical Wire	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9
	less severe <-----> more severe																
Longitudinal	9	8	7	6	5	4	3	2	1	2	3	4	5	6	7	8	9

# QUANTIFY VARIABLES

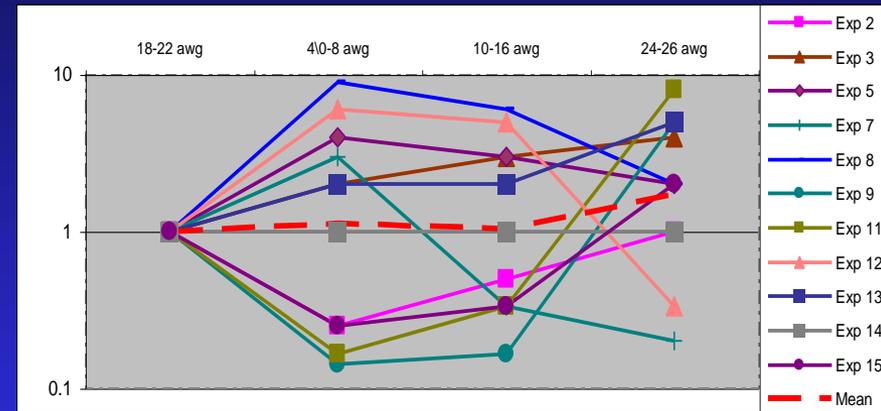
Open Failures

Bundle Protection



Shorting Failures

Wire Gauge



$$\text{geom mean}(y_1, \dots, y_n) = \prod_{i=1}^n y_i^{1/n}$$

Use geometric mean as average expert response

# SELECTION OF PAIRED COMPARISON ENVIRONMENTS

- This selection should
  - ◆ be relatively small
    - ◆ but at a minimum of one plus the number of variables describing the environment
  - ◆ not contain any obviously dominated environments
  - ◆ provides maximum coverage for the regression estimates
  - ◆ contain at least one environment for which failure data exists.
- However, the result should yield a relatively easy paired comparison of the environments

# PAIRED COMPARISON ENVIRONMENTS

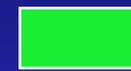
Environment	Wire Gauge	Insulation Type	Conductor Type	Splices	Bundle Size	Bundle Protection	Curvature of Bundle	Bundle Orientation	Ops/Main Traffic	Ops Amplitude	Vibration	Exp Corrosive Fluid	Exp Conducting Fluid	
	1	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Moderate	No	Yes
	2	24-26 awg	Hybrid (PI/FP Composite)	High Streng. Copper Alloy	None	Very Small (< 0.2 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Moderate	No	Yes
	3	24-26 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	Moderate	Benign (P&T Controlled)	Moderate	No	Yes
	4	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Some Level of Prot.	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	High	No	Yes
	5	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Large (> 1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Moderate	Yes	Yes
	6	18-22 awg	Hybrid (PI/FP Composite)	Copper	Non-environmental	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Low	No	Yes
	7	18-22 awg	ETFE & other FPs	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Low	No	Yes
	8	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	High (<= 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Moderate	No	No
	9	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Some Level of Prot.	Low (> 10x)	Horizontal/Vertical Wire	High	D2 (P&T not controlled)	Moderate	No	Yes
	10	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	High (<= 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Low	No	Yes
	11	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Longitudinal	High	Benign (P&T Controlled)	Moderate	No	No
	12	18-22 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	Low	Benign (P&T Controlled)	High	No	Yes
	13	18-22 awg	Polyimide	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	High (<= 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Moderate	No	Yes
	14	4\0-8 awg	Hybrid (PI/FP Composite)	Aluminum	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	Benign (P&T Controlled)	Moderate	No	No
	15	4\0-8 awg	Hybrid (PI/FP Composite)	Copper	None	Moderate (0.5-1.25 in)	Not Protected (Open)	Low (> 10x)	Horizontal/Vertical Wire	High	D2 (P&T not controlled)	Moderate	No	Yes

$$\binom{15}{2} = 105 \text{ comparisons}$$

# PAIRED COMPARISON SURVEY

## Variables

Wire Guage  
 Conductor Type  
 Splices  
 Bundle Protection  
 Curvature of Bundle  
 Ops/Main Traffic  
 Vibration  
 Ops Temp/Presssurization  
 Exp Corrosive Fluid  
 Exp Conducting Fluid  
 Bundle Size  
 Insulation Type  
 Bundle Orientation



Wire Properties



Bundle Properties



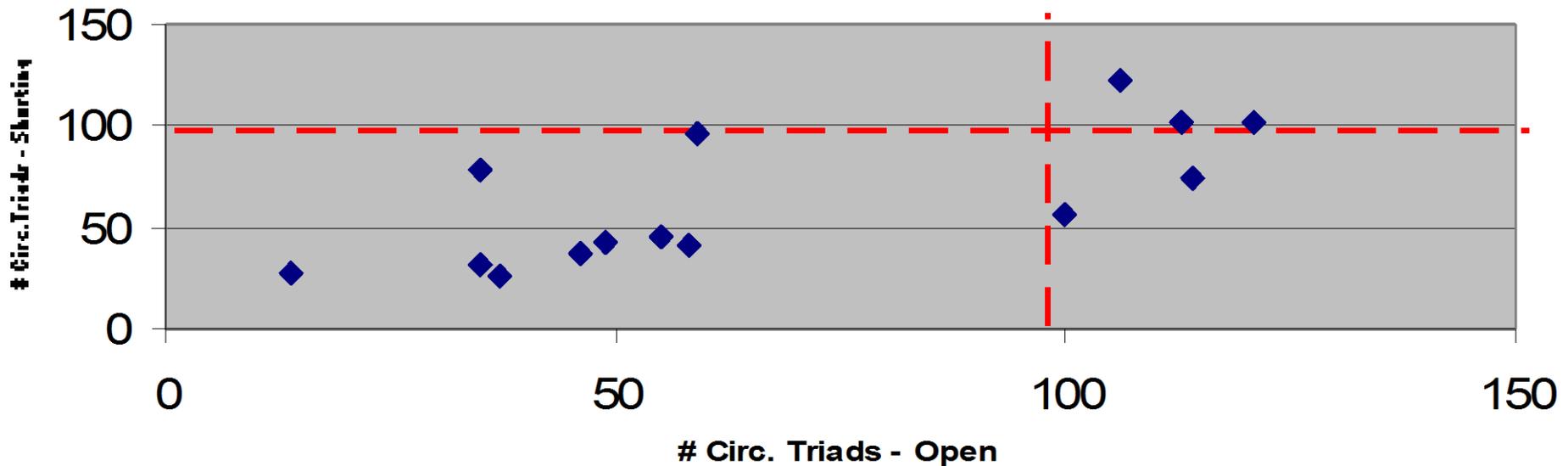
Zonal Properties

# PAIRED COMPARISON SURVEY

WIRE ENVIRONMENT 1		COMPARISON 11	WIRE ENVIRONMENT 2	
<b>WIRE PROPERTIES</b>			<b>WIRE PROPERTIES</b>	
Wire Gauge	18-22 awg		Wire Gauge	
Conductor Type	Copper		Conductor Type	
Insulation Type	Hybrid (PI/FP Composite)		Insulation Type	
Splices	None		Splices	
<b>BUNDLE PROPERTIES</b>			<b>BUNDLE PROPERTIES</b>	
Bundle Size	Moderate (0.5-1.25 in)		Bundle Size	
Bundle Protection	Not Protected (Open)		Bundle Protection	Some Level of Protection
Curvature of Bundle	Low (> 10x)		Curvature of Bundle	
Bundle Orientation (Shock)	Horizontal/Vertical Wire		Bundle Orientation (Shock)	
<b>ZONAL PROPERTIES</b>			<b>ZONAL PROPERTIES</b>	
Ops/Main Traffic	High		Ops/Main Traffic	
Ops Temp/Alt	Benign (P&T Controlled)		Ops Temp/Alt	
Vibration	Moderate		Vibration	High
Exposure to Corrosive Fluid	No		Exposure to Corrosive Fluid	
Exposure to Conductive Fluid	Yes		Exposure to Conductive Fluid	

# PAIRED COMPARISON RESULTS

Plot of Individual Expert Performance



# PAIRED COMPARISON RESULTS

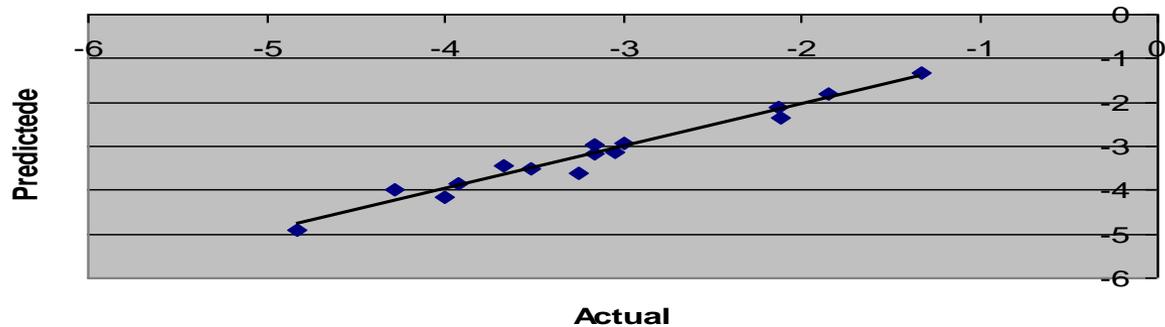
Environemnt	Open Failures			Shorting Failures		
	lower	Bradley-Terry Est	upper	lower	Bradley-Terry Est	upper
1	0.016	0.039	0.068	0.020	0.045	0.067
2	0.060	0.121	0.260	0.047	0.085	0.160
3	0.007	0.026	0.047	0.007	0.019	0.039
4	0.017	0.042	0.073	0.031	0.070	0.130
5	0.068	0.119	0.190	0.077	0.150	0.220
6	0.150	0.265	0.420	0.057	0.102	0.170
7	0.004	0.014	0.029	0.006	0.017	0.032
8	0.021	0.050	0.089	0.012	0.028	0.044
9	0.018	0.042	0.063	0.030	0.059	0.110
10	0.019	0.048	0.080	0.019	0.044	0.075
11	0.004	0.020	0.040	0.003	0.012	0.022
12	0.005	0.018	0.041	0.007	0.024	0.038
13	0.110	0.158	0.260	0.160	0.252	0.430
14	0.001	0.008	0.018	0.004	0.012	0.019
15	0.010	0.030	0.055	0.047	0.081	0.120

# REGRESSION OUTPUT

## SUMMARY OUTPUT OPEN FAILURE ANALYSIS

Regression Statistics					
Multiple R	0.9987				
R Square	0.9975				
Adjusted R Square	0.7929				
Standard Error	0.2868				
Observations	15				
ANOVA					
	df	SS	MS	F	Significance F
Regression	10	161.4031	16.1403	196.2824	0.0001
Residual	5	0.4112	0.0822		
Total	15	161.8142			
	Coefficients	Standard Error	t Stat	P-value	
Intercept	0	#N/A	#N/A	#N/A	
Wire Gauge	0.4535	0.1343	3.3770	0.0197	
Insulation Type	2.0738	0.6439	3.2209	0.0234	
Conductor Type	-0.4380	0.1701	-2.5745	0.0498	
Splices	0.5639	0.0781	7.2246	0.0008	
Curvature of Bundle	0.5013	0.2000	2.5061	0.0541	
Shock Dam. Pot.	-8.1221	0.9121	-8.9051	0.0003	
Ops/Main Traffic	0.2014	0.0560	3.5950	0.0156	
Ops temp/altitude	0.2050	0.1236	1.6585	0.1581	
Vibration	0.2239	0.0924	2.4218	0.0600	
Exp Corrosive Fluid	0.4742	0.1026	4.6237	0.0057	

### Actual vs Predicted Ln(Failure Rate)



## Failure Rate Open Failures

$$\begin{aligned}
 &= \exp\{0 - (-3.1354) + 0.4535 * \text{Wire Gauge Code} \\
 &\quad + 2.0738 * \text{Insulation Type Code} \\
 &\quad - 0.4380 * \text{Conductor Type Code} \\
 &\quad + 0.5639 * \text{Splices Code} \\
 &\quad + 0.5013 * \text{Curvature of Bundle Code} \\
 &\quad - 8.1221 * \text{Shcok Damage Potential Code} \\
 &\quad + 0.2014 * \text{Ops/Main Traffic Code} \\
 &\quad + 0.2050 * \text{Ops Temp/Altitude} \\
 &\quad + 0.2239 * \text{Vibration Code} \\
 &\quad + 0.4742 * \text{Exp Corrosive Fluid Code}\} \\
 &\quad \times 10^{-7} \text{ failures per 100 feet of wire}
 \end{aligned}$$

# CALCULATION OF SCALE CONSTANT

## CABIN LIGHTING WIRING

Report Number	Occurrence Date	Submitter	Operator	Stage of Operation	SDR Type	Report Status	ATA System Code	ATA System	Aircraft Make Name	Aircraft Model Name	Aircraft Series Name	Registration Nbr	Aircraft Serial Nbr
2002021200040	25-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	390AA	27450
2002021200041	25-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	CRUISE	A	CLOSED (	2612	FIRE DET:	BOEING	767	300	386AA	27060
2002021200034	21-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	378AN	25447
2002021200035	21-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	386AA	27060
2002021200027	20-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	390AA	27450
2002011100057	10-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	362AA	24043
2002011000070	07-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	353AA	24034
2002011000072	05-dec-01	AIR CARRIER TAXI	(AMERICAN AIRLINES	INSP/MAINT	A	CLOSED (	3397	LIGHT SY:	BOEING	767	300	386AA	27060

# CALCULATION OF SCALE CONSTANT

Number of Engines	Wing Type	Part Name	Part Condition	Part Location	Nature of Condition	Precautionary Condition
2	MONOPLANE-LOW WING	WIRE	DAMAGE	CABIN	SYSTEM 1	NONE
2	MONOPLANE-LOW WING	WIRE	FALSE INI	LAVATOR	FALSE W/	UNSCHEDED LAN
2	MONOPLANE-LOW WING	WIRE	DAMAGE	CABIN	SYSTEM 1	NONE
2	MONOPLANE-LOW WING	WIRE	DAMAGE	CABIN	SYSTEM 1	NONE
2	MONOPLANE-LOW WING	WIRE	BROKEN	CABIN	SYSTEM 1	NONE
2	MONOPLANE-LOW WING	WIRE	DAMAGE	CABIN	SYSTEM 1	NONE
2	MONOPLANE-LOW WING	WIRE	DAMAGE	CABIN	SYSTEM 1	NONE
2	MONOPLANE-LOW WING	WIRE	BROKEN	CABIN	SYSTEM 1	NONE

# CALCULATION OF SCALE CONSTANT

TURBOFAN/TURBOJET		.....OCTOBER.....		.....NOVEMBER.....		.....DECEMBER.....		.....Quarter.....	
Aircraft Mfg Model	Operator Designator	Number Aircraft	Flight Time	Number Aircraft	Flight Time	Number Aircraft	Flight Time	Average Number Aircraft	Total Flight Time
A A300*	FDRA	43	5,382	43	5,394	41	5,990	43	16,766
Aircraft Model Total:		43	5,382	43	5,394	41	5,990	43	16,766
A A300B4200	DHLA XNDA			6	811	8	1,399	6	811
Aircraft Model Total:				6	811	8	1,399	7	2,110
A A300B4605E	AALA	34	5,909	34	6,409	34	6,852	34	19,170
Aircraft Model Total:		34	5,909	34	6,409	34	6,852	34	19,170
A A300F4605E	ISMA	47	4,250	47	4,875	47	6,248	47	15,373
Aircraft Model Total:		47	4,250	47	4,875	47	6,248	47	15,373
A A310301	FDRA	26	2,141	26	2,240	26	3,209	26	7,590
Aircraft Model Total:		26	2,141	26	2,240	26	3,209	26	7,590
A A310322	FDRA	18	1,489	18	1,711	18	1,798	18	4,998

# OTHER WORKS

Chaloner, K., Church, T., Louis, T. A., and Matts, J.P., (1993), “Graphical elicitation of a prior distribution for a clinical trial”, *Statistician*, 42, pp. 341-355

Zuashkiani, A., Banjevic, D. and Jardine, A.K.S. (2009), “Estimating parameters of the proportional hazards model based on expert knowledge and statistical data”, *Journal of the Operational Research Society*, 60, pp. 1621-1636.